

Ph.D. Course Work Examinations, July-2023

MATHEMATICS

Course-IV-1.4(A): Advanced Fluid Mechanics

Time: 3 Hours

Max. Marks:70

Instructions to Candidates: (i) Answer any Five full questions.

(ii) Each questions carries equal marks.

1. (a) Compute δ_1 , δ_2 and δ_3 when velocity distribution in the boundary layer is

$$u = U \left(\frac{y}{\delta} \right)^n.$$

- (b) Discuss the Blasius-Topppler solution for the boundary layer flow over a flat plate.

(7+7)

2. (a) Discuss the Prandtl's boundary layer theory and what are the importance of Prandtl's boundary layer theory in fluid dynamics?

- (b) What are the main limitations of ideal fluids?

(8+6)

3. (a) Derive Von Karman's integral equation for steady flow under no pressure gradient.

- (b) Based on the Von Karman's integral equation, Determine the local frictional coefficient

C_f , for flow over a flat plate.

(7+7)

4. (a) Explain Momentum integral equation for boundary layer by Von Karman.

- (b) Obtain the Energy integral equation for two-dimensional steady laminar flow of incompressible fluid over a semi-infinite plate.

(7+7)

5. (a) Write short note on (i) Free convection (ii) forced convection (iii) Nusselt number and (iv) Prandtl number.

- (b) Derive Reynold's first integral.

(7+7)

6. (a) Derive thermal boundary layer equations for the flow over a semi-infinite flat plate.

- (b) Explain the solution for cooling problem when the dissipation term is neglected.

(7+7)

7. (a) In a duct in which air is flowing, a normal shock wave occurs at a Mach number 1.5. The static pressure and temperature upstream of the shock are 170 KN/m^2 and 23°C

respectively. Take $\gamma=1.5$ and determine (i) pressure (ii) temperature and (iii) Mach number downstream of the shock and (iv) Strength of the shock.

(b) Discuss Wave equations in two and three dimensions. (7+7)

8. (a) Find the profile $\phi(x, t)$ of a dimensional wave propagation if at $t = 0, \phi = F(x)$ and

$$\frac{\partial \phi}{\partial t} = G(x).$$

(b) Explain Mack number and its importance. (7+7)
